A.

weight of hommer =
$$M_{H}g$$
. It also = $\frac{GM_{H}M_{P}}{V^{2}}$ Planet 20, $M_{H}g = \frac{GM_{H}M_{P}}{V^{2}}$
 $M_{P} = \frac{gV^{2}}{G}$ need g and V .

$$\Delta X = N_{O}t + \frac{1}{2}\alpha t^{2} = \frac{\alpha pply}{\alpha pply} to hammer.$$
 $2.0_{M} = 0 + \frac{1}{2}\alpha (69_{S})^{2}$
 $2 = 2380 \alpha \implies \alpha = \frac{8.403 \times 10^{-4} M_{S}^{2}}{6.3 \times 10^{-4} M_{S}^{2}}$

walk around planet: Circumference = $27TV$
 $6.3 \times 10^{-4} M_{S}^{2} \times 10^{-4} M_{S}^{2} \times 1003 M$
 $M_{P} = \frac{(8.403 \times 10^{-4} M_{S}^{2}) \times 1003 M}{6.672 \times 10^{-11}} = \frac{1.27 \times 10^{13} M_{S}}{1.27 \times 10^{13} M_{S}} M_{S}^{2}$

B. 1. PV = nRT. The question says n and T are the same and R is a universal constant, so PV = a constant. That is, pressure is inversely proportional to volume. The container with three times the volume has one third the pressure: $P_A = (1/3)P_B$.

2.

$$PV = nRT \implies n = \frac{PV}{RT} = \frac{(1.00 \times 10^{-9} \text{ Fe}) (1 \text{ m}^{3})}{(8.314 \frac{J}{\text{mole} \cdot \text{K}}) (300 \text{ k})} = 4.009 \times 10^{-13}$$

$$(4.009 \times 10^{-13} \text{ mole}) \times (6.022 \times 10^{23} \frac{\text{mole} \cdot \text{mole}}{\text{mole}}) = [2.41 \times 10^{11}]$$

$$\text{mole}$$

C. a. The tire's volume stays approximately constant:

$$P_1V = nRT_1$$
 and $P_2V = nRT_2$ $\Longrightarrow \frac{P_2V}{P_1V} = \frac{nRT_2}{nRT_1}$ $\Longrightarrow P_2 = P_1\frac{T_2}{T_1}$

You have to have absolute pressure and temperature: $31 \text{ lb/in}^2 + 14.7 \text{ lb/in}^2 = 45.7 \text{ lb/in}^2$. $45^{\circ}\text{C} + 273 = 318 \text{ K}$. $11^{\circ}\text{C} + 273 = 284 \text{ K}$.

$$P_2 = (45.7) \frac{284}{318} = 40.8 \frac{1b}{in^2}$$

So, the gauge reads
$$40.8 - 14.7 = 26.1 \frac{1b}{in^2}$$

b. This time, the temperature stays constant.

$$P_1V_1 = nRT$$
 and $P_2V_2 = nRT$ \implies $P_1V_1 = P_2V_2$
$$V_2 = \frac{P_1}{P_2}V_1 = \frac{40.8}{14.7}(20 L) = \boxed{55.5 L}$$

D.

$$\alpha = \sqrt[3]{\frac{G(m_1 + m_2)T^2}{4\pi T^2}}$$

$$= \sqrt[3]{\frac{G(m_1 + m_2)T^2}{4\pi T^2}}$$

$$= \sqrt[3]{\frac{(6.67 \times 10^{-11})[2 \times 10^{30} \text{ bg} + (.2)(2 \times 10^{30})][(3 \times 10^{7})][3 \times 10^{7})}{4\pi T^2}}$$

$$= \sqrt[3]{\frac{(6.67 \times 10^{-11})[2 \times 10^{30} \text{ bg} + (.2)(2 \times 10^{30})][(3 \times 10^{7})][3 \times 10^{7})}{4\pi T^2}} = \sqrt[3]{\frac{(6.67 \times 10^{-11})[2.4 \times 10^{30} \text{ bg}][8.987 \times 10^{29}]}{4\pi T^2}} = \sqrt[3]{\frac{3.64 \times 10^{48}}{4\pi T^2}}$$

$$= \sqrt[3]{\frac{(6.67 \times 10^{-11})[2.4 \times 10^{30} \text{ bg}][8.987 \times 10^{29}]}{4\pi T^2}} = \sqrt[3]{\frac{3.64 \times 10^{48}}{3.64 \times 10^{48}}}$$

$$\begin{split} \overline{F}_{64} &= \frac{G \, m_6 m_4}{r_{64}^2} (\widehat{i}) = \frac{(6.672 \, x \, 10^{-11})(6 \, kg)(4 \, kg)}{(4 \, m)^2} (\widehat{i}) \\ &= -10.0 \, x \, 10^{-11} \, \widehat{i} \, N \\ \overline{F}_{24} &= \frac{G \, m_2 m_4}{r_{24}^2} (\widehat{j}) = \frac{(6.672 \, x \, 10^{-11})(2 \, kg)(4 \, kg)}{(3 \, m)^2} (\widehat{j}) \\ &= 5.93 \, x \, 10^{-11} \, \widehat{j} \, N \\ Total force: \overline{F} &= \overline{F}_{64} + \overline{F}_{24} = -100 \, \widehat{i} + 59.3 \, \widehat{j} \, pN \end{split}$$

To find magnitude and direction:

R =
$$\sqrt{100^2 + 59.3^2}$$
 = 116
 θ = arc tan $(\frac{59.3}{100})$ = 31°
$$180 - 31 = 149^{\circ}$$
ans: 116 pN at 149°

F. 1. Because the gravitational force on the satellite points directly toward the Earth (along the \vec{r} vector, making $\theta = 0^{\circ}$ in $\tau = r F \sin \theta$), gravity produces no torque. Therefore, angular momentum is conserved. (Kepler's second law follows from conservation of \vec{L} .)

$$\Gamma_{p}(p/N_{p}) = \Gamma_{A}(p/N_{A}) \qquad P = perigea$$

$$\frac{N_{p}}{N_{A}} = \frac{\Gamma_{A}}{\Gamma_{p}} = \frac{2289 + 6.37 \times 10^{3} \text{ km}}{459 + 6.37 \times 10^{3} \text{ km}} = \frac{8659 \text{ km}}{6829 \text{ km}} \frac{\text{ANS}}{1.27}$$

Add the altitude to the radius of the Earth, from the formula sheet, to get the radius of the orbit. It doesn't matter if you use m or km, but it has to be the same on all distances, so something needs to be converted. The 6762 s period is irrelevant.)

$$T^2 = \frac{4\pi^2}{G(m_1 + m_2)}a^3$$

2. Kepler's third law:

Jupiter is thousands of times as massive as any of its moons, so m_1+m_2 is, to a good approximation, equal to just Jupiter's mass, which I'll call m.

$$m = \frac{4\pi^2}{GT^2}a^3 = \frac{4\pi^2}{(6.67 \times 10^{-11})(1.53 \times 10^5)^2}(4.22 \times 10^8 \, m)^3 = 1.90 \times 10^{27} \, kg$$